## CONFIDENTIAL

SET A



# UNIVERSITI KUALA LUMPUR Malaysia France Institute

# FINAL EXAMINATION

## **SEPTEMBER 2014 SESSION**

SUBJECT CODE	:	FAB30803
SUBJECT TITLE	:	CONTROL SYSTEM
LEVEL	:	BACHELOR
TIME / DURATION	:	9.00 AM – 12.00 PM ( 3 HOURS )
DATE	:	31 DECEMBER 2014

## INSTRUCTIONS TO CANDIDATES

- 1. Please read the instructions given in the question paper CAREFULLY.
- 2. This question paper is printed on both sides of the paper.
- 3. Please write your answers on the answer booklet provided.
- 4. Answer should be written in blue or black ink except for sketching, graphic and illustration.
- 5. This question paper consists of SIX (6) questions. Each question worth 20 marks. Answer FIVE (5) questions only.
- 6. Answer all questions in English.
- 7. The semi-log paper, Nichol's chart, graph paper and formula are appended.

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#### **INSTRUCTION:** Answer FIVE (5) questions only.

#### **Question 1**

(a) Give three (3) objectives of control system design and analysis.

(3 marks)

(b) Draw a block diagram of the components for elevator-position control system.

(5 marks)

(c) The schematic diagram for a speed control system is shown in **Figure 1** in which the DC motor voltage Eg(s) is generated by a DC generator driven by a prime mover. This system has been used on locomotives where the prime mover is a diesel engine that operates most efficiently at one speed. The motor voltage Eg(s) is varied by changing the generator input voltage E(s). Draw all subsystem block diagram and obtain the transfer function  $\frac{\omega(s)}{E(s)}$  of the system. Assume the following for the system

constants:

(12 marks)

- E = speed reference voltage
- $R_f$  = generator field winding resistance
- $L_f$  = generator field winding inductance
- $I_f$  = generator winding current
- $E_g$  = generator output voltage/armature winding input voltage
- $I_a$  = armature winding current
- $R_a$  = armature winding resistance
- $L_a$  = armature winding inductance
- $V_b$  = back emf voltage

- T =output torque
- $\omega$  = output speed
- J = motor and load moment of inertia
- *D* = load and motor viscous friction coefficient
- $K_g$  = generator constant
- $K_b$  = back emf constant
- $K_m$  = motor torque constant



Figure 1: Schematic diagram for a speed control system

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### Question 2

(a) Find the transfer function,  $\frac{C(s)}{R(s)}$  for a block diagram of a system in **Figure 2** by using the block diagram reduction method.

(10 marks)



Figure 2: A block diagram of a system

- (b) **Figure 3** shows a block diagram of a servo system. Answer the following questions:
  - i. Determine the values of gain *K* and velocity feedback constant  $K_h$  so that the maximum overshoot in the unit-step response is 0.2 and the peak time is 1 second. Assume J = 1kg-m<sup>2</sup> and B = 1N-m/rad/sec.

(8 marks)

ii. Determine the settling time  $(T_s)$  for 2% criterion.

(2 marks)



Figure 3: The block diagram of a servo system

#### **Question 3**

(a) **Figure 4** shows a signal flow graph of a system. Obtain the overall transfer function,



Figure 4: A signal flow graph

(b) Referring to a unity feedback system shown in **Figure 5**, find the number of closedloop poles located in the right half-plane, left half-plane, and on the j $\omega$ -axis by using Routh-Hurwitz criterion and comment on the system stability.

(10 marks)



Figure 5: A unity feedback system

### Question 4

(a) Consider the block diagram of a system shown in **Figure 6**.



Figure 6: The block diagram of a system

i. Determine the system type.

(3 marks)

ii. Find the static error constants of  $K_p$ ,  $K_v$  and  $K_a$ .

(3 marks)

iii. Obtain the steady state error for an input of 50u(t), 50tu(t), and  $50t^2u(t)$ .

(3 marks)

(b) Consider a non-unity feedback system shown in **Figure 7**.



Figure 7: A non-unity feedback system

Using the Nyquist criterion, obtain the followings:

i. For *K* =1, sketch the Nyquist plot of  $G(j\omega)$  for  $0 < \omega < \infty$ 

(7 marks)

ii. Determine the range of *K* for stability.

(2 marks)

iii. Obtain the gain margin for K = 300.

(2 marks)

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#### **Question 5**

(a) The block diagram of an electric carrier steering control system is shown in **Figure 8**.



Figure 8: An electric carrier steering control system

i. For K = 3.2 plot the asymptotic Bode plot for the steering control system.

(8 marks)

ii. From the Bode plot, determine the gain margin (*GM*), phase margin (*PM*), gain cross over frequency ( $\omega_{gco}$ ) and phase cross over frequency ( $\omega_{pco}$ ). Hence, draw a conclusion on the stability of the system.

(4 marks)

(b) Complete **Table 1** from the obtained Bode plot in **Question 5(a)i** and plot the locus of  $G(j\omega)$  on a Nichol's chart.

$\omega(rad/s)$	0.5	1.0	1.5	2.0	2.5	3.0	3.5	5.0
$ G(j\omega) (dB)$								
$\angle G(j\omega)(\deg)$								

Table 1

(5 marks)

- (c) From the Nichol's plot in **Question 5(b)**, determine the following for K = 3.2:
  - i. The bandwidth (*BW*) of the closed-loop system.
  - ii. The resonant peak value  $(M_r)$ .
  - iii. The resonance frequency ( $\omega_r$ ).

(3 marks)

#### **Question 6**

Consider a unity feedback system shown in Figure 9.



Figure 9: A unity feedback system

(a) Sketch the Root Locus of the system as *K* varies from zero to infinities. Show all important calculations, i.e. asymptotes, break points on the real axis, imaginary axis crossing points and angle of departure or arrival, where appropriate.

(15 marks)

(b) Determine the closed-loop dominant poles of a system at a damping ratio,  $\zeta$  of 0.4. Indicate the closed-loop poles on the plot and calculate the value of *K* at this point. (5 marks)

### END OF QUESTION

### Appendix 1

1. 
$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta \omega_n s + \omega_n^2}$$

2. %
$$OS = e^{-\left(\zeta \pi / \sqrt{1 - \zeta^2}\right)} \times 100$$

3. 
$$\zeta = \frac{-\ln(\% OS / 100)}{\sqrt{\pi^2 + \ln^2(\% OS / 100)}}$$

4. 
$$T_p = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}}$$

5. 
$$T_s = \frac{4}{\zeta \omega_n}$$
 for 2% criteria

6. 
$$T_r = \frac{\pi/2 + \phi}{\omega_n \sqrt{1 - \zeta^2}}$$

where, 
$$\phi = \tan^{-1} \frac{\zeta}{\sqrt{1-\zeta^2}}$$

7. 
$$e(\infty) = e_{step}(\infty) = \frac{1}{1 + \lim_{s \to 0} G(s)}$$

8. 
$$e(\infty) = e_{ramp}(\infty) = \frac{1}{\lim_{s \to 0} sG(s)}$$

9. 
$$e(\infty) = e_{parabola}(\infty) = \frac{1}{\lim_{s \to 0} s^2 G(s)}$$

10. 
$$K_p = \lim_{s \to 0} G(s)$$

11. 
$$K_{v} = \lim_{s \to 0} sG(s)$$
  
12. 
$$K_{a} = \lim_{s \to 0} s^{2}G(s)$$

## Laplace Transform Table

ltem no.	f(t)	F(s)
1.	$\delta(t)$	1
2.	u(t)	$\frac{1}{s}$
3.	tu(t)	$\frac{1}{s^2}$
4.	$t^n u(t)$	$\frac{n!}{s^{n+1}}$
5.	$e^{-at}u(t)$	$\frac{1}{s+a}$
6.	$\sin \omega t u(t)$	$\frac{\omega}{s^2+\omega^2}$
7.	$\cos \omega t u(t)$	$\frac{s}{s^2 + \omega^2}$